

NEW ZEALAND CIVIL AIRWORTHINESS REQUIREMENTS

VOLUME 2

AIRWORTHINESS DIRECTIVES

SECTION A

BRITISH AEROSPACE DOVE AD SCHEDULE

In addition to the ADs scheduled below, relevant ADs in the 'AIRCRAFT GENERAL' AD Schedule and any other relevant ADs in Section C 'COMPONENTS AND EQUIPMENT' shall be complied with

DCA/DH104/1	Mandatory Modifications
Applicability:	All model 104
Requirement:	Embody all applicable UK CAA mandatory modifications
Compliance:	Prior to initial C of A issue
DCA/DH104/2	Centre Section and Wing - Safe Life
Applicability:	All model 104
Requirement:	Replace centre section and wing lower spar boom components per Hawker Siddeley TNS 119
Compliance:	As detailed in TNS 119
Effective Date:	19 July 1971
DCA/DH104/3	Wing Assembly - Inspection
Applicability:	All model 104 in which mod 852 was not embodied during construction
Requirement:	Inspect wing spar booms for corrosion per Hawker Siddeley TNS 125
Compliance:	As detailed in TNS 125
Effective Date:	26 June 1967
DCA/DH104/4	Wing Spar Booms - Inspection
Applicability:	Model 104 up to S/N 04465 without mod. 870 embodied
Requirement:	X-ray per Hawker Siddeley TNS 134
Compliance:	As detailed in TNS 134
Effective Date:	17 March 1975

DCA/DH104/5	Wing Lower Root Joint - Inspection
Applicability:	All model 104
Requirement:	Inspect fittings per Hawker Siddeley TNS 168
Compliance:	At intervals not exceeding 4500 hours TIS or 6 years, whichever is the sooner
Effective Date:	12 July 1971
DCA/DH104/6	Control Column - Inspection
Applicability:	All model 104
Requirement:	Inspect per Hawker Siddeley TNS 171
Compliance:	Whenever aileron control circuit in column is disturbed
Effective Date:	11 July 1960
DCA/DH104/7	Trim Tab Controls - Inspection
Applicability:	All model 104
Requirement:	Inspect rudder and elevator connecting rod eye ends P/N CM 2A and CM 28 per Hawker Siddeley TNS 187
Compliance:	At intervals not exceeding 2400 hours TIS
Effective Date:	24 February 1964
DCA/DH104/8	Engine Mounts - Inspection
Applicability:	All model 104 with engine mounts P/N 4 EM 201A and 203A
Requirement:	X-ray per Hawker Siddeley TNS 190
Compliance:	At intervals not exceeding 4 years
Effective Date:	12 January 1970
DCA/DH104/9	Rudder Control System - Inspection
Applicability:	All model 104
Requirement:	Inspect reversal lever P/N 4CF 767A per Hawker Siddeley TNS 219
Compliance:	At intervals not exceeding 2400 hours TIS
Effective Date:	19 October 1970
DCA/DH104/10	Engine Mount Attachment - Inspection
Applicability:	All model 104
Requirement:	Inspect attachment fittings per Hawker Siddeley TNS 226
Compliance:	As detailed in TNS 226
Effective Date:	25 October 1971

DCA/DH104/11	Engine Mount - Inspection
Applicability:	All model 104
Requirement:	Inspect LH mounting frame P/N 4EM 201A per Hawker Siddeley TNS 228
Compliance:	At intervals not exceeding 75 hours TIS
Effective Date:	17 April 1972
DCA/DH104/12	Nose Undercarriage - Inspection
Applicability:	All model 104
Requirement:	Inspect inner casing per Hawker Siddeley TNS 232
Compliance:	At intervals not exceeding 600 hours TIS until mod. 1728 embodied and thereafter at intervals not exceeding 2400 hours TIS
Effective Date:	17 September 1973
DCA/DH104/13	Centre Section - Inspection
Applicability:	All model 104
Requirement:	Inspect centre section main spar top boom P/N 4FS. 135 A/1 per Hawker Siddeley TNS 237
Compliance:	As detailed in TNS 237
Effective Date:	9 September 1976
DCA/DH104/14	Centre Section - Inspection
Applicability:	All model 104 without mod. 779 embodied
Requirement:	Inspect main spar bottom boom per Hawker Siddeley TNS 238
Compliance:	As detailed in TNS 238
Effective Date:	24 January 1977
DCA/DH104/15	Aircraft Life Extension - Modification
Applicability:	All model 104
Requirement:	Modify per British Aerospace TNS 240
Compliance:	When mod. 1707 is embodied for extension of aircraft life
DCA/DH104/16	Engine Mount - Inspection
Applicability:	All model 104
Requirement:	Inspect front engine support spigot P/N 4EM 681 per British Aerospace TNS 244
Compliance:	By 30 November 1978

DCA/DH104/17**Aileron Balance Weight Assembly - Inspection****Applicability:**

All model 104

Requirement:

Accomplish visual and x-ray inspections per British Aerospace TNS 245 Issue 2.

Assemblies with internal or external corrosion exceeding stated limits must be removed from service before further flight.

*Note: Radiographs are to be sent to British Aerospace for interpretation as prescribed in TNS 245 Issue 2 and results notified to CAD when received***Compliance:**

Visual inspection - within next 14 days

X-ray inspection - within next 50 hours TIS or 28 days, whichever is the sooner

Effective Date:

25 June 1987

DCA/DH104/18**Flap Jack to Flap Attachment - Inspection****Applicability:**

All model 104

Requirement:

To prevent possible asymmetric flap operation due to installation of an unapproved, non-safety locked pin, visually inspect each flap installation to ensure that correct pin P/N 4 WF 329 is fitted and correctly locked.

British Aerospace TNS 246 refers

Compliance:

By 26 December 1988

Effective Date:

19 December 1988

*Note: 1. Requirement notified to registered owners on effective date.
2. Copy of reference document may be obtained from the Director.*